

CONDITION-BASED MAINTENANCE OF COMPLEX TECHNICAL SYSTEMS

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This paper examines condition-based maintenance (CBM) of complex technical systems using aircraft engines as an example. CBM requires continuous vibration monitoring to predict the optimal moment for scheduled maintenance. At any given moment, an engine may exist in multiple possible states, only one of which is realized. This set of possible states represents the physical potential of the system. To determine the actual state, this study proposes using Harrington's desirability function, which establishes a relationship between linguistic assessments of an engine's technical condition and normalized numerical intervals.

KEYWORDS:

Aircraft engine, condition-based maintenance, complex technical system, vibration monitoring, Harrington's function, linguistic assessment.

1 INTRODUCTION

Modern aircraft engines are characterized by high standards of design and manufacturing, ensuring stable performance over extended periods. Advances in technology have enabled engines to be equipped with monitoring sensors that provide objective data on their operational condition [Zaborowski 2007, Olejarova 2017, Chaus 2018, Matiskova 2021]. These developments facilitate a shift from scheduled maintenance to condition-based maintenance, optimizing operating costs and improving aircraft reliability [Kostochkin 1988, Monkova 2013, Balara 2018, Pollak 2020, Duhancik 2024].

One of the key indicators providing objective information about engine condition is vibration level. Vibration analysis helps detect dynamic changes preceding failures. This study applies the theory of dynamic systems operating in a "blow-up mode" to predict engine lifespan based on vibration analysis [Baron 2016, Murcinkova 2018, Flegner 2019 & 2020]. This approach provides accurate information about an engine's actual condition and enables highly precise maintenance predictions [Straka 2018a,b].

Decision-making regarding the criticality of an engine's technical condition is formalized using a desirability function. This function correlates linguistic evaluations of technical parameters in physical scales with numerical intervals of the desirability function, allowing for efficient condition-based maintenance [Michalik 2014, Nekrasov 2017 & 2020, Kuchar 2018, Harnicarova 2019].

The demonstration of the effectiveness of this approach is the primary objective of the present study.

2 RESEARCH METHODOLOGY

In dynamic systems developing in a blow-up mode, a periodic process is superimposed on the primary trend of a monitored parameter. This process can be described by a mathematical model [Vagaska 2017 & 2021], where one of the coefficients corresponds to the moment of system failure or a radical change in its operating conditions [Nagornyi 2016, Panda 2014 & 2020, Pandova 2020, Sukhodub 2018 & 2019, Nahornyi 2022].

Such modes are described by the following equation:

$$\frac{dx}{dt} = x^{1+1/\alpha}. \quad (1)$$

The solution of this equation increases indefinitely as it approaches the critical moment t_{f_f} :

$$x(t) \sim (t_f - t)^{-\alpha}. \quad (2)$$

To obtain a practical solution, we transition from the real indicator aa to a complex one, leading to the following equation:

$$x(t) = \text{Re} \sum_k a_k (t_f - t)^{-\alpha + k\beta i} = (t_f - t)^{-\alpha} \cdot F(\ln(t_f - t)). \quad (3)$$

The function $F(\cdot)$ is represented by multiple harmonics, characterizing the system's nonlinear behavior in blow-up mode. However, in practical applications [Panda, 2024], the function is often approximated by its first harmonic:

$$x(t) = (t_f - t)^{-\alpha} \cdot \left(a_0 + a_1 \cos \left(\beta \cdot \ln \frac{t_f - t}{\tau} \right) \right). \quad (4)$$

This expression defines a smooth trend over which log-periodic oscillations are superimposed. These oscillations act as precursors, signaling an approaching blow-up moment t_f . As

$t \rightarrow t_f$ approaches, the oscillation frequency increases indefinitely, satisfying the dynamic law governing blow-up mode. The continuous rise in log-periodic oscillation frequency allows for early detection of catastrophic processes long before t_f .

If we interpret the exhaustion of engine life T as the blow-up moment t_f , then engine can be classified as operating in blow-up mode.

To improve engine life prediction accuracy, it is essential to isolate the sensitive log-periodic component of the recorded signal [Panda 2021 & 2022, Kurdel 2022]. This requires separating the total periodic signal from the smooth trend and analyzing it independently during the operational period [Zakhakhatnov 2022].

The periodic component model should undergo direct analysis, fully capturing the complex polyharmonic structure of the recorded signal [Mrkvica 2012, Labun 2018 & 2020].

The monitored parameter $A_{CON}(t)$ is expressed as the sum of a smooth trend B_{TR} and a periodic component A_{PER} :

$$A_{CON}(t) = B_{TR}(t) + A_{PER}(t). \quad (5)$$

The trend component B_{TR} is determined using:

$$B_{TR}(t) = A_0 + b \cdot (T - t)^\alpha. \quad (6)$$

The periodic component A_{PER} is derived from:

$$A_{PER}(t) = A_1 \cos(\omega \cdot \ln(T-t) - \varphi),$$

where

$$A_1 = C \cdot (T-t)^{-\alpha}; \quad \varphi = \pi - \omega \cdot \ln(T-t).$$

Equations (6) and (7) contain four unknown parameters $A_0, \omega, \varphi, \alpha$, which are computed using a specialized computer program.

The program minimizes the difference (8) between the experimentally determined time series $A_{CON}(t)$ and the model expressions (6) and (7):

$$\sum_i^m (A_{CON}(t_i) - B_{TR}(t_i) - A_{PER}(t_i))^2 \Rightarrow \min. \quad (8)$$

In practice, predicting the remaining operational lifespan T_{RUL} is of key interest:

$$T_{RUL} = T - t. \quad (9)$$

Minimization of function (8) is performed stepwise within an expanding time window, moving from left to right. The window expands until the available experimental data is exhausted. The number of forecasted T_{RUL} values forms a predictive time series, with the number of steps equal to the time axis steps t_i . Once the forecasted operational hours T before maintenance is determined, decision-making regarding the criticality of the engine's condition is formalized using the Harrington desirability function.

This function correlates linguistic assessments of an engine's technical condition with normalized numerical intervals d (Tab.1).

Table 1. Harrington's Desirability Scale Intervals

Linguistic Assessment	Desirability Function Interval d
Excellent	1.00 - 0.80
Normal	0.80 - 0.63
Acceptable	0.63 - 0.37
Degraded	0.37 - 0.00

Harrington's desirability function is given by:

$$d = \exp\left(-\exp\left(-\frac{t_i}{T}\right)\right). \quad (10)$$

3 EXPERIMENTAL SECTION

During the operational period between scheduled maintenance, forecasts are based on analyzing engine vibration along the Y-axis (Fig. 1) [Doroshko 1984].

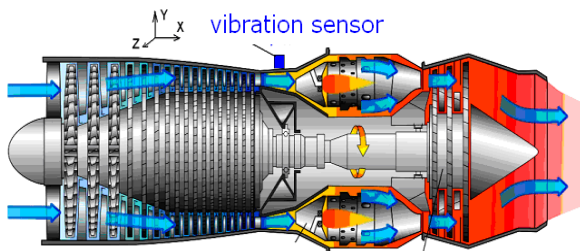


Figure 1. Sensor placement on the engine

Approximation of the recorded vibration signal A_{CON} using the predictive model and the point of scheduled maintenance interruption is shown in Fig. 2.

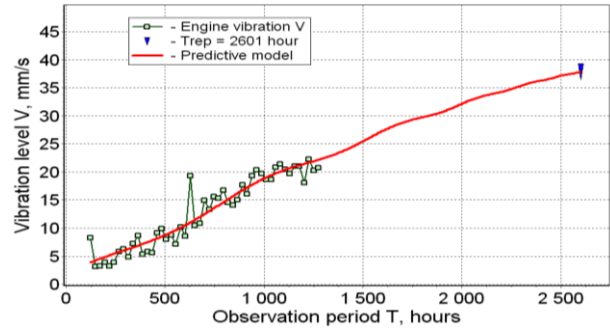


Figure 2. Approximation of experimental data using the predictive model

Changes in Harrington's desirability function d during engine operation are shown in Fig. 3.

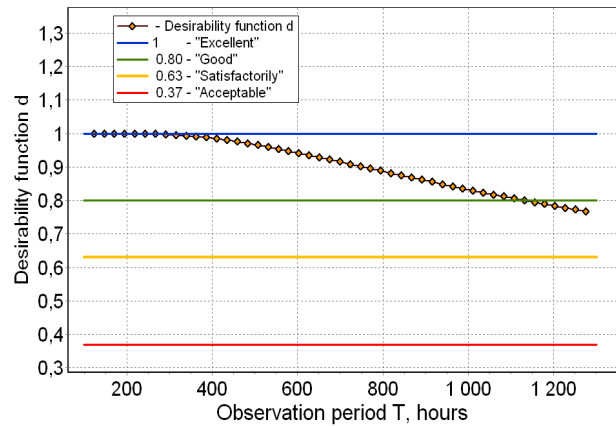


Figure 3. Changes in Harrington's function d

The remaining operational lifespan T_{RUL} over time is demonstrated in Fig. 4.

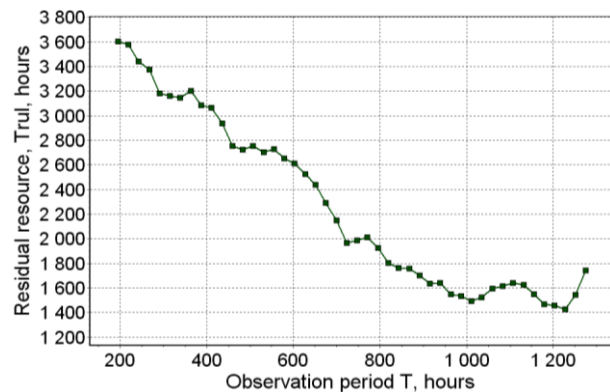


Figure 4. Changes in the remaining engine lifespan T_{RUL}

Table 2. Predictive protocol

Flight Time Before Forecast:	1275.00 hours
Forecasted flight time before maintenance:	2601.26 hours
Confidence interval ($P = 0.95$):	2600.46 - 2602.05 hours

During the observation period (Fig. 3), the engine's condition was classified as "Excellent" according to the desirability function. At 1275 operational hours, the forecast indicated a remaining lifespan of approximately 1400 hours (Fig. 4). The predictive protocol (Tab. 2) estimated a total operational time of around 2600 hours before maintenance.

4 CONCLUSIONS

The high level of design, manufacturing, and operation of aircraft engines ensures long-term stable performance. Advances in monitoring technology provide objective data for condition-based maintenance. By treating aircraft engines as dynamic systems developing in a blow-up mode, it becomes possible to mathematically predict operational hours before scheduled maintenance. Harrington's desirability function enables formalization of decision-making regarding engine condition, establishing a correlation between linguistic assessments and normalized numerical intervals. This approach facilitates the transition from scheduled to condition-based maintenance, optimizing operating costs and improving aircraft reliability.

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